

AIRWORTHINESS DIRECTIVE

| ENGINE – Valve Push Rod Assembly – INSPECTION/REPLACEMENT | | |
|---|---|--|
| AD No: | 2018-0001 | |
| Issue Date: | 11 May 2018 | |
| Effective Date: | 11 May 2018 | |
| Type Approval Holder Name: | | Type/Model Designation(s): |
| BRP-POWER | TRAIN GmbH & Co. KG | Rotax 912 and 914 engines |
| Related Mandato | ry Continuing Airworthiness I | nformation Reference: |
| Rotax SB-912i-0 Rotax SB-912-0 | 017-0208, dated 13 October 20 008R1, dated 12 October 2017 070R1, dated 12 October 2017 052R1, dated 12 October 2017 | 17 |
| Supersedure | | |
| N/A | | |
| Related EASA Se | rvice Information Bulletin refe | ence: |
| N/A | | |
| Manufacturer(s): | | |
| BRP-Powertrain | n GmbH & Co. KG | |
| Applicability: | | |
| | s Rotax 912UL, 912ULS, 914UL en affected refer to Rotax SB SB-912 i | gines. i-008iS/SB-912-070UL / SB914-052UL at latest revision |
| Justification: | | |
| determined that assemblies mar | , due to a quality control deficiency | ted on Rotax 912/914 engines in service. It has been in the manufacturing process of certain valve push-rod and 02 October 2017 inclusive, partial wear on the rocker unction of the valve train. |
| | | ead to rough engine operation and loss of power, possibly age to the aeroplane and injury to occupants. |
| To address this potential unsafe condition, BRP-Rotax issued Service Bulletin (SB) SB-912 i-008 / SB-912- 070 / SB-914-052 (single document), providing applicable instructions. | | |

For the reason described above, this AD requires a one-time inspection and, depending on findings, replacement of affected parts.

This AD also prohibits installation of affected parts on an engine. This AD, for the uncertified engine variants, mirrors the EASA AD 2017-0208, which was for the certified engine variants.

Compliance:

Required as indicated unless accomplished previously:

Note 1: Valve push-rod assemblies Part Number (P/N) 854861 are hereafter referred to as "valve push-rod" in this AD.

Note 2: BRP Rotax SB-912 i-008 / SB-912-070 / SB-914-052 (single document) is hereafter referred to as "the SB" in this AD.

Note 3: BRP Rotax SB-912 i-008iS / SB-912-070UL / SB-914-052UL (single document) is hereafter referred to as "the Applicability SB" in this AD.

Note 4: For the purpose of this AD, Group 1 engines are those having a serial number (s/n) as listed in the Applicability SB (see Note 3 of this AD); or an engine, having any s/n, on which a valve push-rod (see Note 1 of this AD) has been replaced in service between 08 June 2016 and the effective date of this AD inclusive. Group 2 engines are those that are not Group 1.

Inspection:

(1) For Group 1 engines (see Note 4 of this AD): Within the compliance time identified in Table 1 of this AD, as applicable, visually inspect the push-rod ball sockets of each valve push-rod in accordance with the instructions of the SB.

Table 1 – Visual Inspection of Affected Assembly

| Engine Flight Hours (FH) since first installation on an aircraft | Compliance Time |
|---|--|
| 160 FH or less | Before exceeding 170 FH since first installation of the engine on an aircraft, or within 3 months after the effective date of this AD, whichever occurs first. |
| More than 160 FH | Within 10 FH or 3 months, whichever occurs first after the effective date of this AD |

Corrective action:

(2) If, during the inspection as required by paragraph (1) of this AD, a valve push-rod having black surface is detected (reference Fig. 1 in the SB), before next flight, replace that valve push-rod and its affected parts (see Note 5 of this AD) with serviceable ones in accordance with the instructions of the SB.

Note 5: for the purpose of this AD, an affected part is listed in Table 2 of this AD.

Table 2 – Affected Part

| Part | Part Number |
|-------------------------|-------------|
| Valve push-rod assembly | 854861 |
| Rocker arm left | 854383 |
| Rocker arm right | 854393 |

Part installation:

(3) For Group 1 and Group 2 engines: From the effective date of this AD, it is prohibited to install on any engine a valve push-rod (see Note 1 of this AD) manufactured between 08 June 2016 and 02 October 2017 inclusive

Reference Publications:

BRP-Rotax SB-912 i-008iS / SB-912-070UL / SB-914-052UL (single document) Initial Issue 12 October 2017

BRP-Rotax SB-912 i-008 R1 / SB-912-070 R1 / SB-914-052 R1 (Single Document) Revision 1 12 October 2017

The use of later approved revisions of this document is acceptable for compliance with the requirements of this AD

Remarks:

- 1. If requested and appropriately substantiated, TM CAD can approve Alternative Methods of Compliance for this AD.
- 2. Enquiries regarding this AD should be referred to the Airworthiness Inspectorate, Transport Malta Civil Aviation Directorate. E-mail: <u>civil.aviation@transport.gov.mt</u>.